Milestone Review Flysheet

PDR, CDR, FRR

| Institution Name | Massachusetts Institute of Technology | |
|------------------|---------------------------------------|---------|
| Institution Name | Massachusetts Institute of Tech | ınology |

| Milestone CDR |
|---------------|
|---------------|

| Vehicle Properties | | |
|---------------------------------|---------------------|--|
| Diameter (in) 6.28 | | |
| Length (in) 124.5 | | |
| Gross Liftoff Weight (lb) 49.27 | | |
| Launch Lug/button Size | 3/16" | |
| Motor Retention | Stainless Steel Rod | |

| Stability Analysis | | |
|-----------------------------------|------------------|--|
| Center of Pressure (in from nose) | 89.95 | |
| Center of Gravity (in from nose) | 77.09 | |
| Static Stability Margin | 2.11 | |
| Thrust-to-Weight Ratio | 8.3 | |
| Rail Size (in) / Length (in) | 1.5" x 1.5", 8ft | |

| Recovery System Properties | | | | | |
|----------------------------------|------------------|--|----------------------------|-----------|--|
| | Drogue Parachute | | | | |
| Manufactu | ırer/Model | Sur | plus Military Pilot | | |
| Si | ze | | 36" | | |
| Altituo | le at Deploym | ent (ft) | 5,650 | | |
| Velocit | y at Deployme | ent (ft/s) | ~20 ft/s depending on wind | | |
| Terminal Velocity (ft/s) | | 46.35 | | | |
| Recovery Harness Material | | Tubuluar Nylon | | | |
| Harness Size/Thickness (in) | | 1" | | | |
| Recovery Harness Length (ft) | | 16' | | | |
| Harness/Airframe of payload sa | | a 3/8" forged eye bolt in top abot unit. Eye bolt also bebbing that goes to tender | | | |
| Kinetic Energy During Descent | Section 1 | Section 2 | Section 3 | Section 4 | |
| (ft-lb) | 1320 | | | | |

| Recovery System Properties | | |
|---|--|--|
| Electronics/Ejection | | |
| Altimeter(s) Make/Model | Featherweight Raven2 | |
| Redundancy Plan | Perfectflite Stratologger, redundant drogue ejection charges. Redundant igniters in Tender Descender on main | |
| Pad Stay Time (Launch Configuration) | Upwards of 8 hours | |

| Motor Properties | | | |
|------------------------------|---------------|--|--|
| Motor Manufacturer Cesaroni | | | |
| Motor Designation | L1115 | | |
| Max/Average Thrust (N/lb) | 1713N, 1119N | | |
| Total Impulse (N-sec/lb-sec) | 5015 Ns | | |
| Mass pre/post Burn (lb) | 4.4kg, 2.01kg | | |

| Ascent Analysis | | |
|---------------------------|-------|--|
| Rail Exit Velocity (ft/s) | 54 | |
| Max Velocity (ft/s) | 594.6 | |
| Max Mach Number | 0.53 | |
| Max Acceleration (ft/s^2) | 503.3 | |
| Peak Altitude (ft) | 5,015 | |

| Recovery System Properties | | | | | |
|--------------------------------|---------------|---------------|------------------------------------|-----------|--|
| Main Parachute | | | | | |
| Manufactu | ırer/Model | | RocketMan | | |
| Si | ze | | 8' | | |
| Altitud | de at Deploym | ent (ft) | 500' | | |
| Velocit | y at Deployme | ent (ft/s) | 46 | .35 | |
| Lan | ding Velocity | (ft/s) | 13.9 | | |
| Recovery Harness M | | Iaterial | Tubular Nylon | | |
| Harness Size/Thickne | | ess (in) | 1" | | |
| Recovery Harness Length (f | | ngth (ft) | 3.25' | | |
| TT /4 * C | | bay, connecte | ye-nut into top d to threaded r | | |
| Kinetic Energy Upon Landing | Section 1 | Section 2 | Section 3 | Section 4 | |
| (ft-lb) | 70.7 (body) | .8 (Nose/Sabo | 9.7 (Quadroto | r) | |

| Recovery System Properties | | |
|---|---------------------------------------|--|
| Electronics/Ejection | | |
| Rocket Locators (Make, Model) | Big Red Bee 70cm Transmitter/2m GPS. | |
| Transmitting Frequencies | 2.4GHz 6 channels 915 Mhz | |
| Black Power Mass | 3.5 | |
| Drogue Parachute (gram) | | |
| Black Power Mass Main Parachute (gram) | 0.2 (Tender Descender release device) | |

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| | | | |

| Payload/Science | | |
|--|--|--|
| Succinct Overview of Payload/Science Experiment | A Custom built composite quadrotor carrying a system of payloads to exhibit object tracking and recognition in dynamic environments and to quantitatively measure high-altitude atmospheric lightning events | |
| Identify Major Components | The primary payload of the rocket will be the quadrotor which will be equipped with GPS, radio telecommand devices, an IMU, pressure sensors, an onboard computes, CMOS cameras, IR emitters/detectors, and an autonomous flight stabilizer. | |
| Mass of Payload/Science | The mass of the quadrotor is being estimated to be about 10 pounds. | |

| Test Plan Schedule/Status | | |
|---------------------------|--------------------------|--|
| Ejection Charge Test(s) | 12/14/ 2012 (Successful) | |
| Sub-scale Test Flights | 12/15/2012 (Successful) | |
| Full-scale Test Flights | Early February | |

| Additional Comments | | |
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